Flow in Pipe

Introduction

Study of viscosity on an incompressible, internal flow.

E.g., Flow in a circular pipe

$$\operatorname{Re} = \frac{V\rho l}{\mu}$$

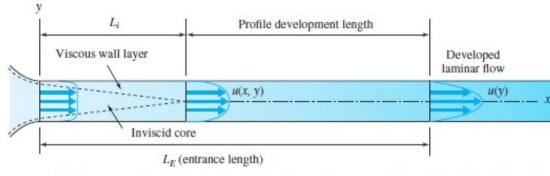
- The Reynolds number is a ratio of inertial forces to viscous forces.
 - Important when dealing with viscous effects in a flow.
- When the ratio is large, inertial forces dominate viscous forces.
 - True when there are short, sudden geometric changes.
 - Viscous effects are important when surface areas are large.

Introduction

Re =
$$\frac{V\rho l}{\mu}$$

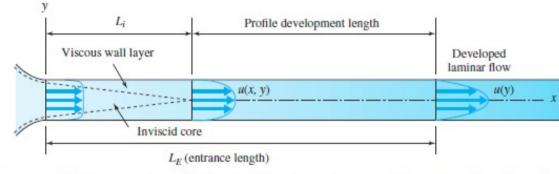
Laminar Flow

- Re < 2000 for pipes
- Re < 1500 in a wide channel
- At a sufficiently high Reynolds number, a turbulent flow occurs.



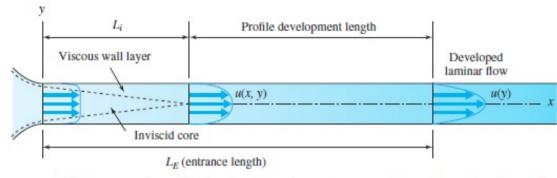
Entrance region of a laminar flow in a pipe or a wide rectangular channel.

- Developed Laminar Flow Flow where the velocity profile ceases to change in the flow direction.
- In the entrance region of a laminar flow, the velocity profile changes in the flow direction.
- Idealized flow from a reservoir begins at the inlet as a uniform flow.
- Viscous wall layer grows over the inviscid core length, L_i until the viscous stresses dominate the entire cross section.



Entrance region of a laminar flow in a pipe or a wide rectangular channel.

- The profile develops due to viscous effects until a developed flow is achieved.
 - The inviscid core length is one-fourth to one-third of the entrance length L_E.
 - This depends on the conduit geometry, shape of the inlet velocity profile, and the Reynolds number.



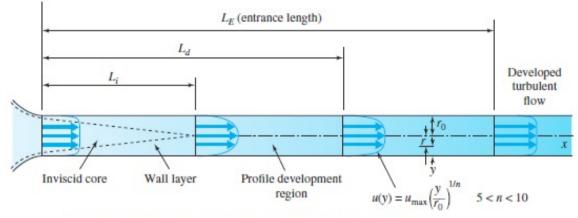
Entrance region of a laminar flow in a pipe or a wide rectangular channel.

$$\frac{L_E}{D} = 0.065 \,\mathrm{Re} \qquad \mathrm{Re} = \frac{VD}{V}$$

The entrance length equation for a **laminar flow in a circular pipe** with a uniform profile at the inlet. $Re_{lam} = 2000$

$$\frac{L_E}{h} = 0.04 \,\mathrm{Re} \qquad \mathrm{Re} = \frac{Vh}{v}$$

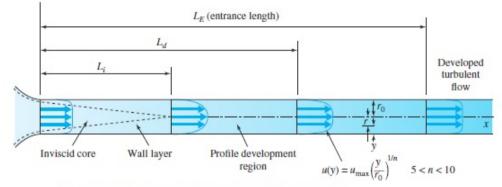
For a **laminar flow in a highaspect-ratio channel** with a uniform profile at the inlet. Re_{lam} = 1500



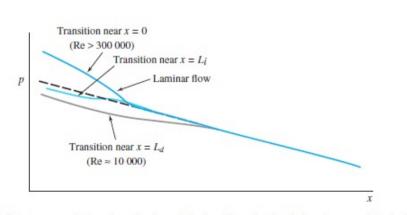
Velocity profile development in a turbulent pipe flow.

For a large Reynolds number (Re > 10⁵)

$$\frac{L_i}{D} \simeq 10$$
 $\frac{L_d}{D} \simeq 40$ $\frac{L_E}{D} \simeq 120$



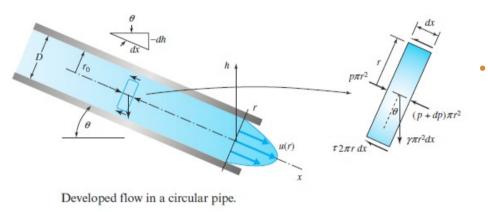




Pressure variation in a horizontal pipe flow for both laminar and turbulent flows. (From PhD Thesis of Dr. Jack Backus, Michigan State University)

- For a flow beyond a large x, the pressure variation decreases linearly with x.
- Transition near origin → Linear pressure variation begins near L_i → Pressure gradient in the inlet is higher than in the developed flow region.
- Transition near L_d→ (Low Re)→ Linear variation begins at the end of the transition→ Pressure gradient in the inlet is less than that of developed flow.
- Laminar flow→ Pressure variation is the same as that from a large Reynolds number→ Pressure gradient is higher than in the developed flow region.

Elemental Approach

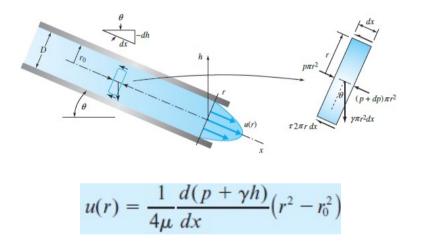


To investigate incompressible, steady, developed laminar flow in a pipe

Elemental approach:

- Infinitesimal control volume into which and from which fluid flows [Use momentum equation].
- Infinitesimal fluid mass upon which forces act [Use Newton's second law].
- Since velocity profile doesn't change in the x-direction:
 - Momentum Flux in = Momentum Flux out and the resultant force is zero
 - No acceleration of the mass element; resultant force is zero.

Elemental Approach



- The velocity distribution is **parabolic**.
- Called a **Poiseuille flow**.
 - This is a laminar flow with a parabolic profile in a pipe or between parallel plates.

Solving the Navier-Stokes Equations

streamlines
steady steady
$$\rho\left(\frac{\partial u}{\partial t} + p_r \frac{\partial u}{\partial r} + \frac{\partial u}{r \partial \theta} \frac{\partial u}{\partial \theta} + u \frac{\partial u}{\partial x}\right)^{\text{flow}}$$

$$= -\frac{\partial p}{\partial x} + \gamma \sin \theta + \mu \left(\frac{\partial^2 u}{\partial r^2} + \frac{1}{r \partial r} \frac{\partial u}{\partial r} + \frac{1}{r^2} \frac{\partial^2 u}{\partial \theta^2} + \frac{\partial^2 u}{\partial x^2}\right)^{\text{symmetric developed}}$$

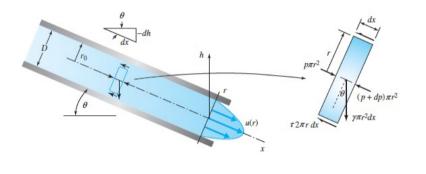
$$u(r) = \frac{\lambda}{4} \left(r^2 - r_0^2\right)$$
$$= \frac{1}{4\mu} \frac{d(p+\gamma h)}{dx} \left(r^2 - r_0^2\right)$$

Navier-Stokes equation for flow in a circular pipe:

- Developed flow
- Streamlines are parallel to the wall
- No swirl
- No acceleration of the fluid particles as they move in the pipe.

This **parabolic** velocity distribution for flow in a pipe is called a **Poiseuille flow**.

Pipe Flow Quantities



For steady, laminar, developed flow in a circular pipe, the velocity distribution is:

$$u(r) = \frac{1}{4\mu} \frac{d(p+\gamma h)}{dx} \left(r^2 - r_0^2\right)$$

• The average velocity, V, is:

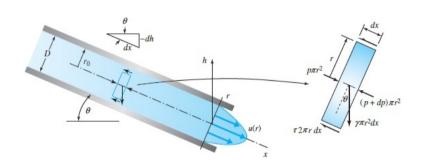
$$V = \frac{Q}{A} = \frac{1}{\pi r_0^2} \int_0^{r_0} u(r) \ 2\pi r \ dr$$
$$= \frac{2}{r_0^2} \int_0^{r_0} \frac{1}{4\mu} \frac{d(p+\gamma h)}{dx} (r^2 - r_0^2) r \ dr = -\frac{r_0^2}{8\mu} \frac{d(p+\gamma h)}{dx}$$

• For a horizontal pipe, the pressure drop is as follows:

$$\Delta p = \frac{8\mu VL}{r_0^2}$$
$$\Delta p = \frac{2\tau_0 L}{r_0}$$

For an inclined pipe, p is replaced with $(p+\gamma h)$

Pipe Flow Quantities



- The friction factor, f :
 - Dimensionless wall shear valid for both laminar and turbulent flow.

$$f = \frac{\tau_0}{\frac{1}{8}\rho V^2}$$

$$\begin{array}{c} \frac{64}{\text{Re}} & \text{For laminar flow} \\ \text{in a pipe} \end{array}$$

• Also, head loss (h_L) with a dimension of length:

$$\frac{\Delta p}{\gamma} = h_L = f \frac{L}{D} \frac{V^2}{2g}$$

Darcy-Weisbach equation (valid for both laminar and turbulent flows).

$$h_L = \frac{32\mu LV}{\gamma D^2}$$

Head-loss is directly proportional to the average velocity in a laminar flow.

A small-diameter horizontal tube is connected to a supply reservoir as shown in Figure E7.1. If 6600 mm³ is captured at the outlet in 10 s, estimate the viscosity of the water.



Solution

The tube is very small, so we expect viscous effects to limit the velocity to a small value. Using Bernoulli's equation from the surface to the entrance to the tube, and neglecting the velocity head, we have, letting 0 be a point on the reservoir surface,

$$\frac{p_h^{\bullet}}{p_y^{\bullet}} + H = \frac{V_{\perp}^{\bullet}}{2g} + \frac{p}{\gamma}$$

where we have used gage pressure with $p_0 = 0$. This becomes, assuming $V^2/2g \approx 0$ at the tube's entrance,

 $p = \gamma H = 9800 \text{ N/m}^2 \times 2 \text{ m} = 19600 \text{ Pa}$

At the exit of the tube the pressure is zero; hence

$$\frac{\Delta p}{L} = \frac{19\,600}{1.2} = 16\,300 \text{ Pa/m} (\text{N/m}^3)$$

The average velocity is found to be

$$V = \frac{Q}{A} = \frac{6600 \times 10^{-9}/10}{\pi \times 0.001^2/4} = 0.840 \text{ m/s}$$

Check to make sure the velocity head is negligible: $V^2/2g = 0.036$ m compared with $p/\gamma = 2$ m, so the assumption of negligible velocity head is valid and our pressure calculation is acceptable. Using Eq. 7.3.14, we can find the viscosity of this assumed laminar flow to be

$$\mu = \frac{r_0^2}{8V} \frac{\Delta p}{L} = \frac{0.0005^2 \,\mathrm{m}^2}{8 \times 0.84 \,\mathrm{m/s}} (16\,300 \,\mathrm{N/m^3}) = \frac{6.06 \times 10^{-4} \,\mathrm{N \cdot s/m^2}}{10^{-4} \,\mathrm{N \cdot s/m^2}}$$

We should check the Reynolds number to determine if our assumption of a laminar flow is acceptable. It is

$$\operatorname{Re} = \frac{\rho VD}{\mu} = \frac{1000 \text{ kg/m}^3 \times 0.84 \text{ m/s} \times 0.001 \text{ m}}{6.06 \times 10^{-4} \text{ N} \cdot \text{s/m}^2} = 1390$$

where we use kg/m³ = $N \cdot s^2/m^4$. This is obviously a laminar flow since Re < 2000, so the calculations are valid providing the entrance length is not too long. It is

$$L_E = 0.068 \,\mathrm{Re} \times D = 0.065 \times 1390 \times 0.001 = 0.09 \,\mathrm{m}$$

This is approximately 8% of the total length, a sufficiently small quantity; hence the calculation for viscosity is assumed acceptable.

Derive an expression for the velocity distribution and the flow rate between horizontal, concentric pipes for a steady, incompressible developed flow (Figure E7.2).

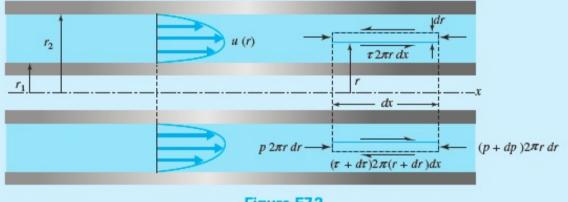


Figure E7.2

Solution

Let us use an elemental approach. The element is a hollow cylindrical shell as sketched in the figure. If we sum forces, we obtain

$$p2\pi r \, dr - (p+dp)2\pi r \, dr + \tau 2\pi r \, dx - (\tau+d\tau)2\pi (r+dr)dx = 0$$

Simplifying, there results, neglecting the term of differential magnitude,

$$\frac{dp}{dx} = -\frac{\tau}{r} - \frac{d\tau}{dr} - \frac{d\tau}{r}$$

Substituting $\tau = -\mu \ du/dr$ (du/dr is negative near the outer wall where the element is sketched) we have

$$\frac{dp}{dx} = \mu \left(\frac{1}{r} \frac{du}{dr} + \frac{d^2 u}{dr^2} \right)$$
$$= \frac{\mu}{r} \frac{d}{dr} \left(r \frac{du}{dr} \right)$$

Multiply both sides by *rdr* and divide by μ , then integrate:

$$r\frac{du}{dr} = \frac{1}{2\mu}\frac{dp}{dx}r^2 + A$$

Multiply both sides by *dr/r* and integrate again:

$$u(r) = \frac{1}{4\mu} \frac{dp}{dx}r^2 + A \ln r + B$$

where A and B are arbitrary constants. They are found by setting u = 0 at $r = r_1$ and at $r = r_2$; that is,

$$0 = \frac{1}{4\mu} \frac{dp}{dx} r_1^2 + A \ln r_1 + B$$
$$0 = \frac{1}{4\mu} \frac{dp}{dx} r_2^2 + A \ln r_2 + B$$

Solve for A and B:

$$A = \frac{1}{4\mu} \frac{dp}{dx} \frac{r_{1}^{2} - r_{2}^{2}}{\ln(r_{2}/r_{1})}$$
$$B = -A \ln r_{2} - \frac{r_{2}^{2}}{4\mu} \frac{dp}{dx}$$

Thus

$$u(r) = \frac{1}{4\mu} \frac{dp}{dx} \left[r^2 - r_2^2 + \frac{r_2^2 - r_1^2}{\ln(r_1/r_2)} \ln(r/r_2) \right]$$

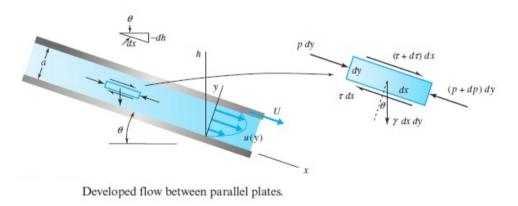
This is integrated to give the flow rate:

$$Q = \int_{r_1}^{r_2} u(r) 2\pi r \, dr$$
$$= -\frac{\pi}{8\mu} \frac{dp}{dx} \left[r_2^4 - r_1^4 - \frac{\left(r_2^2 - r_1^2\right)^2}{\ln(r_2/r_1)} \right]$$

As $r_1 \rightarrow 0$ the velocity distribution approaches the parabolic distribution of pipe flow. As $r_1 \rightarrow r_2$ this distribution approaches that of parallel-plate flow. These two conclusions are not obvious and are presented as Problem 7.48 at the end of this chapter.

For incompressible, steady, developed flow of a fluid between parallel plates, with the upper plate moving with velocity U.

Elemental Approach



- For an elemental volume of unit depth (in the z-direction)
 - One dimensional flow, no acceleration, developed flow.

$$u(y) = \frac{1}{2\mu} \frac{d(p+\gamma h)}{dx} (y^2 - ay) + \frac{U}{a} y$$

Elemental Approach

$$u(y) = \frac{1}{2\mu} \frac{d(p+\gamma h)}{dx} (y^2 - ay) + \frac{U}{a} y$$

- **Couette Flow**: A flow with a **linear profile** resulting from the motion of the plate only.
- **Poiseuille Flow**: If the motion is only due to the pressure gradient (with U = 0).

Solving the Navier-Stokes Equations

- For a developed flow between parallel plates:
 - Streamlines are parallel to the plates so u = u(y); v = w = 0.

$$\rho\left(\frac{\partial \mu}{\partial t} + u\frac{\partial \mu}{\partial x} + \frac{\partial \mu}{\partial y} + \frac{\partial u}{\partial z}\right)$$

$$= -\frac{\partial \rho}{\partial x} + \mu\left(\frac{\partial^2 \mu}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} + \frac{\partial^2 \mu}{\partial z^2}\right) + \gamma \sin\theta$$
developed wide channel

$$u(y) = \frac{\lambda}{2}(y^2 - ay) + \frac{U}{a}y$$
$$= \frac{1}{2\mu}\frac{d(p + \gamma h)}{dx}(y^2 - ay) + \frac{U}{a}y$$

 $\frac{\partial^2 u}{\partial y^2} = \frac{1}{\mu} \frac{d(p+\gamma h)}{dx}$

This analysis applies to the midsection away from the side-walls

Double-integrating with u = 0, y = 0; u = U;y = a.

Simplified Flow Situation

• Velocity distribution between fixed plates (U = 0) is:

Flow rate per unit width:

$$u(y) = \frac{1}{2\mu} \frac{d(p + \gamma h)}{dx} (y^{2} - ay)$$

$$Q = \int u \, dA$$

$$= \int_{0}^{a} \frac{1}{2\mu} \frac{d(p + \gamma h)}{dx} (y^{2} - ay) \, dy = -\frac{a^{3}}{12\mu} \frac{d(p + \gamma h)}{dx}$$
Average velocity:

$$V = \frac{Q}{a \times 1}$$

$$= -\frac{a^{2}}{12\mu} \frac{d(p + \gamma h)}{dx}$$

Simplified Flow Situation

• Velocity distribution between fixed plates (U = 0) is: $u(y) = \frac{1}{2\mu} \frac{d(p + \gamma h)}{dx} (y^2 - ay)$

The pressure drop in terms of average velocity (horizontal channel):

• For plates on an incline, p is replaced with $(p + \gamma h)$

The maximum velocity occurs at y = 0.5a and is:
$$u_{max} = -\frac{a^2}{8\mu} \frac{dy}{dz}$$

Hence, average and maximum velocities are related by:

 $V = \frac{2}{3}u_{\max}$

The pressure drop Δp over a length L of horizontal channel is:

$$\Delta p = \frac{2\tau_0}{a}L \qquad \qquad \tau_0 = -\frac{a}{2}\frac{dp}{dx}$$

$$\Delta p = \frac{12\mu VL}{a^2}$$

Simplified Flow Situation

• Friction factor, f: Dimensionless wall shear valid for both laminar and turbulent flow.

$$f = \frac{\tau_0}{\frac{1}{8}\rho V^2}$$

The pressure drop in terms of friction factor is:

$$\frac{\Delta p}{\gamma} = f \frac{L}{2a} \frac{V^2}{2g}$$

$$f = \frac{8}{\rho V^2} \left(-\frac{a}{2} \frac{dp}{dx} \right) = \frac{8}{\rho V^2} \left(-\frac{a}{2} \right) \left(-\frac{12\mu V}{a^2} \right) = \frac{48\mu}{\rho a V} = \frac{48}{\text{Re}}$$

• Head Loss: Pressure drop due to friction as fluid flows through a pipe.

$$h_L = \frac{12\mu LV}{ra^2}$$

Using definitions for Reynolds number and friction factor; seen that head loss is directly proportional to average velocity.

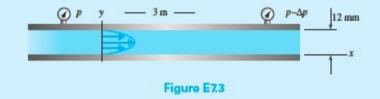
Water at 20°C flows with a Reynolds number of 1500 between the 500-mm-wide, horizontal plates shown in Figure E7.3. Calculate (a) the flow rate,

(b) the wall shear stress,

(b) the wan shear stress,

(c) the pressure drop over 3 m, and

(d) the velocity at y = 0.5 cm



Solution

Since the Reynolds number is 1500, the laminar flow equations are assumed applicable. (a) Using the definition of the Reynolds number, the average velocity is found as follows:

$$1500 = \frac{Va}{v}$$

$$\therefore V = \frac{1500v}{a} = \frac{1500 \times 10^{-6}}{1.2 \times 10^{-2}} = 0.14 \text{ m/}$$

Thus

 $Q = AV = 0.125 \times 0.12 \times 0.5 = 7.5 \times 10^{-4} \text{ m}^{3}\text{/s}$

(b) Using Eq. 7.4.17, the pressure drop over length L is

$$\frac{\Delta p}{L} = \frac{12\mu V}{a^2} = \frac{12 \times 10^{-3} \times 0.125}{0.012^2} = 10.4 \text{ Pa/m}$$

The shearing stress at the wall is found, using Eq. 7.4.22, to be

$$\tau_0 = \frac{a}{2} \frac{\Delta p}{L} = \frac{0.012}{2} \times 10.4 = \underline{0.0624 \text{ Pa}}$$

(c) The pressure drop over 3 m is found to be

 $\Delta p = 10.4L = 10.4 \times 3 = 31.2 \text{ Pa}$

(d) The velocity distribution of Eq. 7.4.14 is

$$u(y) = \frac{1}{2\mu} \frac{dp}{dx} (y^2 - ay)$$

= $\frac{1}{2 \times 10^{-3}} (-10.4) (y^2 - 0.012y) = -5200(y^2 - 0.012y)$

where we have used $dp/dx = -\Delta p/L$. At y = 5 mm, the velocity is

 $u = -5200 (0.005^2 - 0.012 \times 0.005) = 0.182 \text{ m/s}$

We have used three significant digits since the fluid properties are assumed known to three significant digits.

Find an expression for the pressure gradient between two parallel plates that results in a zero shear stress at the lower wall, where y = 0; also, sketch the velocity profiles for a top plate speed of U with various pressure gradients. Assume horizontal plates.

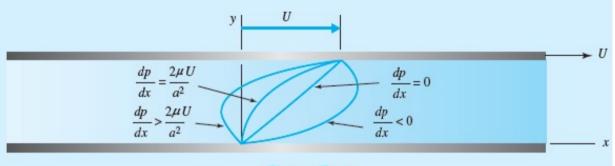


Figure E7.4

Solution

The velocity distribution for plates with the top plate moving with velocity U is given by Eq. 7.4.17. Letting dh/dx = 0, we have

$$u(y) = \frac{1}{2\mu} \frac{dp}{dx} (y^2 - ay) + \frac{U}{a} y$$

The shear stress is

$$\tau = \mu \frac{du}{dy}$$
$$= \frac{1}{2} \frac{dp}{dx} (2y - a) + \mu \frac{U}{a}$$

If $\tau = 0$ at y = 0, then du/dy = 0 at y = 0 and the pressure gradient is

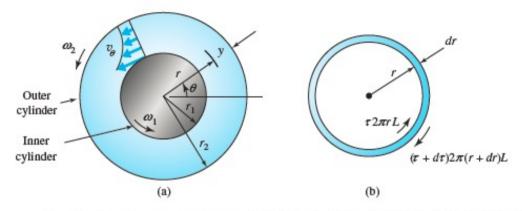
$$\frac{dp}{dx} = \frac{2\mu U}{a^2}$$

If dp/dx is greater than this value, the slope du/dy at y = 0 is negative and thus the velocity u will be negative near y = 0. If dp/dx = 0, we observe that a linear velocity distribution results, namely,

$$u(y) = \frac{U}{a}y$$

If dp/dx is negative, u(y) is greater at each y-location than the linear distribution since $(y^2 - ay)$ is a negative quantity for all y's of interest.

All of the results above can be qualitatively displayed on a sketch of u(y) for several dp/dx as shown in Figure E7.4.



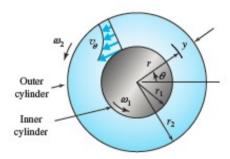
Flow between concentric vertical cylinders: (a) basic flow variables; (b) element from between the cylinders.

- Fully developed, steady flow between concentric, rotating cylinders.
 - Laminar flow is valid up to Re = 1700
 - Above this, a secondary laminar flow may develop, and eventually a turbulent flow develops.

Elemental Approach

- Neglecting body forces (vertical cylinder).
- Pressure doesn't vary with θ; resultant torque acting on an element (thin cylindrical shell) is zero as there is no angular acceleration.

 $\tau 2\pi r L \times r - (\tau + d\tau) 2\pi (r + dr) L \times (r + dr) = 0 \qquad \qquad \blacktriangleright \qquad 2\tau + r \frac{d\tau}{dr} = 0$



 $v_{\theta}(r) = \frac{A}{2}r + \frac{B}{r}$

With the constants to be found by evaluating the boundary conditions:

$$v_{\theta} = r_1 \omega_1$$
 at $r = r_1$, and $v_{\theta} = r_2 \omega_2$ at $r = r_2$

$$A = 2\frac{\omega_2 r_2^2 - \omega_1 r_1^2}{r_2^2 - r_1^2} \qquad B = \frac{r_1^2 r_2^2 (\omega_1 - \omega_2)}{r_2^2 - r_1^2}$$

Solving the Navier-Stokes Equations

- Steady, laminar flow between concentric cylinders.
 - Hence, circular streamlines $v_r = v_z = 0$, $v_\theta = v_\theta(r)$ only, and $\partial p/\partial \theta = 0$

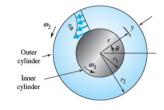
$$\begin{array}{c} \begin{array}{c} \begin{array}{c} \text{symmetric} \\ \left(\frac{\partial \overline{\psi}_{\theta}}{\partial t} + \frac{\partial}{\rho_{r}} \frac{\partial v_{\theta}}{\partial r} + \frac{v_{\theta}}{r} \frac{\partial \overline{\psi}_{\theta}}{\partial \theta} + v_{z} \frac{\partial \overline{\rho}_{\theta}}{\partial z} + \frac{\partial}{\rho_{r}} v_{\theta}}{\partial z} \right) \\ = -\frac{1}{r} \frac{\partial p}{\partial \theta} + \mu \left[\frac{\partial^{2} v_{\theta}}{\partial r^{2}} + \frac{1}{r} \frac{\partial v_{\theta}}{\partial r} + \frac{1}{r^{2}} \frac{\partial^{2} \overline{\psi}_{\theta}}{\partial \theta^{2}} + \frac{\partial^{2} \overline{\psi}_{\theta}}{\partial z^{2}} + \frac{2}{r^{2}} \frac{\partial \overline{\psi}_{\theta}}{\partial \theta} - \frac{v_{\theta}}{r^{2}} \right] \\ = 0 \end{array}$$

$$\begin{array}{c} \text{symmetric} \\ \text{symmetric} \\ \text{flow} \end{array} \quad \text{symmetric} \\ \text{symmetric} \\ \text{flow} \end{array} \quad \text{symmetric} \\ \text{flow} \end{array}$$

Cancelling the terms and double-integration leads to the same equations as in the previous slide. $r^{2}\omega = r^{2}\omega \qquad r^{2}r^{2}(\omega = \omega)$

$$A = 2 \frac{r_2^2 \omega_2 - r_1^2 \omega_1}{r_2^2 - r_1^2} \qquad B = \frac{r_1^2 r_2^2 (\omega_1 - \omega_2)}{r_2^2 - r_1^2}$$

Flow with the Outer Cylinder Fixed ($\omega_2 = 0$)



• E.g., For a shaft rotating in a bearing.

Velocity Distribution:

$$v_{\theta} = \frac{r_1^2 \omega_1}{r_2^2 - r_1^2} \left(\frac{r_2^2}{r} - r \right)$$

Shearing stress:

$$\tau_{1} = -\left[\mu r \frac{d}{dr} \left(\frac{v_{\theta}}{r}\right)\right]_{r=\eta}$$
$$= \mu \frac{2}{r_{1}^{2}} \frac{r_{1}^{2} r_{2}^{2} \omega_{1}}{r_{2}^{2} - r_{1}^{2}} = \frac{2\mu r_{2}^{2} \omega_{1}}{r_{2}^{2} - r_{1}^{2}}$$

Torque, T to rotate the inner cylinder of length, L:

$$T = \tau_1 A_1 r_1$$

$$T = \frac{2\mu r_2^2 \omega_1}{r_2^2 - r_1^2} 2\pi r_1 L r_1 = \frac{4\pi \mu r_1^2 r_2^2 L \omega_1}{r_2^2 - r_1^2}$$

Power to rotate the shaft (multiply torque by rotational speed):

$$\vec{W} = T\omega_1 = \frac{4\pi\mu r_1^2 r_2^2 L \omega_1^2}{r_2^2 - r_1^2}$$

Power needed to overcome resistance of viscosity \rightarrow Leads to an increase in internal energy and temperature of the fluid.

Estimate the viscosity of an oil contained in the annulus between two 25-cm-long cylinders, as shown in Figure E7.6. The outer stationary cylinder is 80 mm in diameter. The 78-mm-diameter inner cylinder rotates at 3800 rpm when a torque of $1.2 \text{ N} \cdot \text{m}$ is applied. The specific gravity of the oil is 0.85. Neglect any torque due to the cylinder ends.

Solution

Assuming that the Reynolds number is less than 1700, Eq. 7.5.19 provides

$$\mu = \frac{T(r_2^2 - r_1^2)}{4\pi r_1^2 r_2^2 L \omega_1}$$

= $\frac{1.2 \text{ N} \cdot \text{m}(0.04^2 - 0.039^2) \text{ m}^2}{4\pi \times 0.04^2 \text{ m}^2 \times 0.039^2 \text{ m}^2 \times 0.25 \text{ m} \times (3800 \times 2\pi/60) \text{ rad/s}} = \underline{0.0312 \text{ N} \cdot \text{s/m}^2}$

Check the Reynolds number using $v = \mu/\rho$:

$$\operatorname{Re} = \frac{\omega_{1} r_{1} \delta}{v} = \frac{(3800 \times 2\pi/60) \operatorname{rad/s} \times 0.039 \operatorname{m} \times 0.002/2 \operatorname{m}}{0.0312/(1000 \times 0.85) \operatorname{m}^{2}/\mathrm{s}} = 423$$

This is less than 1700 so the calculation is acceptable.

Show that as the inner cylinder radius of Figure E7.6 approaches the outer cylinder radius the velocity distribution approaches the linear distribution between parallel plates with one plate moving and a zero pressure gradient. This is Couette flow.

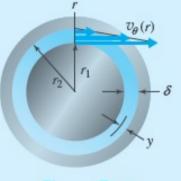


Figure E7.6

Solution

For this problem we will let $\omega_2 = 0$; the velocity distribution (7.5.17) is

$$\begin{aligned} v_{\theta}(r) &= \frac{r_1^2 \omega_1}{r_2^2 - r_1^2} \left(\frac{r_2^2}{r} - r \right) \\ &= \frac{r_1^2 \omega_1}{r_2^2 - r_1^2} \frac{r_2^2 - r^2}{r} = \frac{r_1^2 \omega_1}{r_2^2 - r_1^2} (r_2 - r) \frac{r_2 + r}{r} \end{aligned}$$

Introduce the independent variable y, measured from the outer cylinder defined by $r + y = r_2$ (see Figure 7.6); let $\delta = r_2 - r_1$. Then the above can be written as

$$v_{\theta}(r) = \frac{r_1^2 \omega_1 (r_2 - r)}{(r_2 - r_1)(r_2 + r_1)} \frac{r_2 + r_1}{r}$$
$$= \frac{r_1^2 \omega_1 y}{\delta(r_2 + r_1)} \frac{2r_2 - y}{r_2 - y}$$

As the inner radius approaches the outer radius we can write $r_1 \simeq r_2$. Letting $r_2 \simeq r_1 \simeq R$ we have $r_2 + r_1 \simeq 2R$ and

$$\frac{2R - y}{R - y} \simeq 2$$

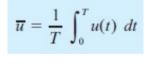
since $y \ll R$. The velocity distribution then simplifies to

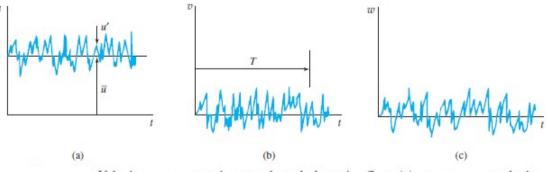
$$v_{\theta}(y) = \frac{R^2 \omega_1 y}{\delta 2R} \times 2 = \frac{R \omega_1}{\delta} y$$

This is a linear distribution and is a good approximation to the flow whenever $\delta \ll R$.

Turbulent Flow in a Pipe

- Developed turbulent flow in a circular pipe is of interest in practical applications (most flows in pipes are turbulent).
 - Laminar flows have been seen in Reynolds numbers of 40,000 in a pipe flow.
 - In standard conditions, Re_{turb} = 2000.
- All three velocity components are nonzero.
 - Need time-average quantities.





Velocity components in a steady turbulent pipe flow: (a) x-component velocity u; (b) r-component velocity v; (c) θ -component velocity w.

Show that $\overline{u'} = 0$ and $\frac{\overline{\partial u}}{\partial y} = \frac{\partial \overline{u}}{\partial y}$ for a turbulent flow.

Solution

To show that $\overline{u'} = 0$ we simply substitute the expression (7.6.1) for u(t) into Eq. 7.6.2 and obtain

$$\overline{u} = \frac{1}{T} \int_0^T (\overline{u} + u') dt$$
$$= \frac{1}{T} \int_0^T \overline{u} dt + \frac{1}{T} \int_0^T u' dt$$
$$= \overline{u} \frac{1}{T} \int_0^T dt + \overline{u'}$$
$$= \overline{u} + \overline{u'}$$

Subtracting \overline{u} from both sides results in

 $\overline{u'} = 0$

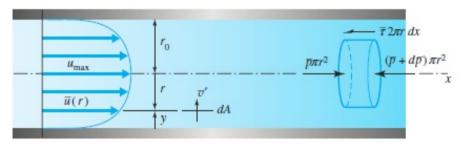
Now, let us time average the derivative $\partial u/\partial y$. We have

$$\begin{aligned} \overline{\frac{\partial u}{\partial y}} &= \frac{1}{T} \int_0^T \frac{\partial u}{\partial y} dt \\ &= \frac{\partial}{\partial y} \left(\frac{1}{T} \int_0^T u \, dt \right) = \frac{\partial}{\partial y} \overline{u} \end{aligned}$$

since T is a constant. Thus

$$\frac{\overline{\partial u}}{\partial y} = \frac{\partial \overline{u}}{\partial y}$$

Differential Equation



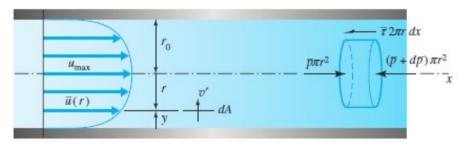
Turbulent flow in a horizontal pipe.

The differential x-force from the random motion of a fluid particle through an incremental area dA is:
 u' is the negative change

$$dF = -\rho v' \, dA \, u'$$

- u' is the negative change in xcomponent velocity.
- The turbulent shear stress is: $\tau_{turb} = \frac{dF}{dA} = -\rho u'v'$
- Time-average turbulent shear stress is the *apparent* shear stress. $\overline{\tau}_{turb} = -\rho \overline{u'v'}$

Differential Equation



Turbulent flow in a horizontal pipe.

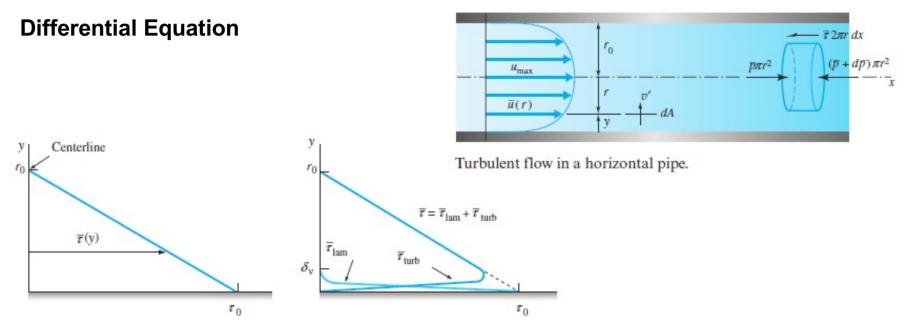
• The total shear stress (from viscosity and momentum exchange) due to laminar and turbulent effects would be:

$$\overline{\tau} = \overline{\tau}_{\text{lam}} + \overline{\tau}_{\text{turb}}$$

$$= \mu \frac{\partial \overline{u}}{\partial y} - \rho \overline{u'v'}$$

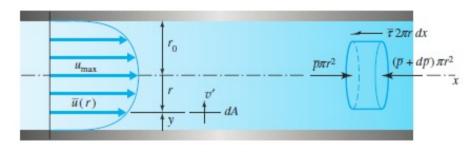
• The shear stress related to the pressure gradient is:

$$\overline{\tau} = -\frac{r}{2}\frac{d\overline{p}}{dx} = \frac{r\Delta\overline{p}}{2L}$$



- Shear stress distribution is linear for turbulent and laminar flow.
- Turbulent shear goes to zero at the wall.
- Total shear at the centerline is zero.

Differential Equation



Turbulent flow in a horizontal pipe.

 To find the time-average velocity distribution, the differential equation is formed from equations on the previous slide.

$$\frac{r}{2}\frac{d\overline{p}}{dx} = \rho \overline{u'v'} + \mu \frac{d\overline{u}}{dr}$$

The term $\overline{u'v'}$ cannot be determined analytically.

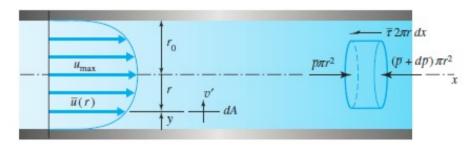
 Have Eddy viscosity: (Parameterization of Eddy Momentum Flux, Reynolds stresses)

$$\overline{u'v'} = \eta \, \frac{d\overline{u}}{dy}$$

• Hence:

$$\frac{r}{2}\frac{d\overline{p}}{dx} = \rho(\nu + \eta)\frac{d\overline{u}}{dr}$$

Differential Equation



Turbulent flow in a horizontal pipe.

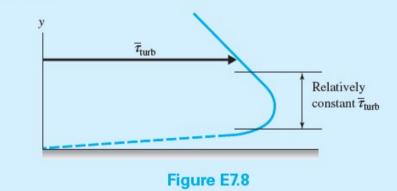
- For turbulent flow, it is helpful to define a **mixing length I**_m:
 - Distance a particle moves before interacting with another particle.

$$\eta = l_m^2 \frac{d\overline{u}}{dy}$$

- The correlation coefficient K_{uv} is a normalized turbulent shear stress.
 - Has limits of ±1
 - With time-averaged quantities.

$$K_{uv} = \frac{\overline{u'v'}}{\sqrt{u'^2}\sqrt{v'^2}}$$

Note that in Figure 7.9b there is a region near the wall where the turbulent shear is near its maximum and is relatively constant, as shown in the expanded view of Figure E7.8, and the viscous shear is quite small. Assume that the mixing length is directly proportional to the distance from the wall. With this assumption show that $\overline{u}(y)$ is logarithmic in this region near the wall.



Solution

If the viscous shear is negligible (as it is away from the thin wall layer), we have, combining Eqs. 7.6.5 and 7.6.8, and 7.6.10,

$$\overline{\tau}_{\rm turb} = \rho \eta \frac{d\overline{u}}{dy} = \rho l_m^2 \left(\frac{d\overline{u}}{dy}\right)^2$$

Now, if $\overline{\tau}_{uvb} = \text{const.} = c_1$ and we assume, as given in the problem statement, that

```
l_m = c_2 y
```

there results

$$c_1 = \rho c_2^2 y^2 \left(\frac{du}{dy}\right)^2$$

or

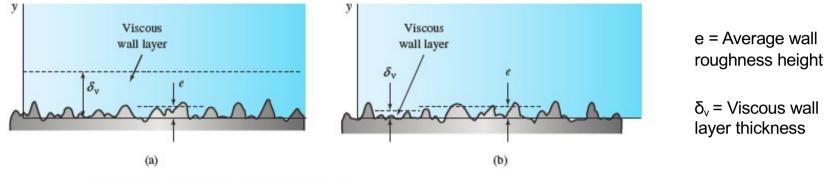
$$y\frac{d\overline{u}}{dy} = c_3$$

where $c_3 = \sqrt{c_1/\rho c_2^2}$. This is integrated to yield

 $\overline{u}(y) = c_3 \ln y + c_4$

Hence, with the foregoing assumptions we see that a logarithmic profile is predicted for the region of constant turbulent shear near the wall. This is, in fact, observed from experimental data; so we conclude that the above assumptions are reasonable for a turbulent flow in a pipe.

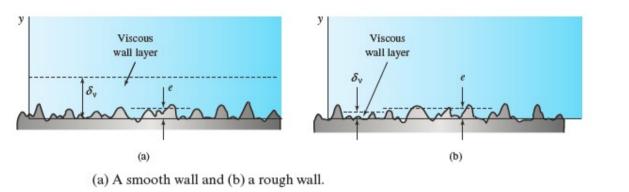




(a) A smooth wall and (b) a rough wall.

- Hydraulically smooth: The viscous wall thickness (δ_v) is large enough that it submerges the wall roughness elements → Negligible effect on the flow (almost as if the wall is smooth).
- If the viscous wall layer is very thin → Roughness elements protrude off the layer → The wall is rough.
- The relative roughness e/D and Reynolds number can be used to find if a pipe is smooth/rough.

Velocity Profile



e = Average wall roughness height

 δ_v = Viscous wall layer thickness

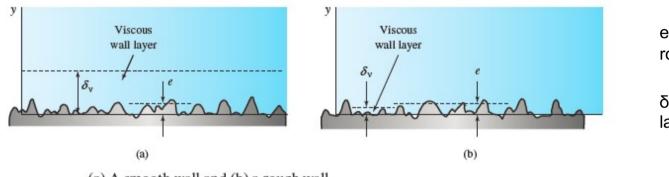
- For a smooth wall, there are two regions of flow (wall and outer regions).
- Wall region: Characteristic velocity = shear velocity $u_{\tau} = \sqrt{\frac{\tau_0}{\rho}}$; Characteristic length = viscous length $\frac{v}{u_{\tau}}$

$$\frac{\overline{u}}{u_{r}} = \frac{u_{\tau}y}{v} \text{ (viscous wall layer)} \quad 0 \le \frac{u_{\tau}y}{v} \le 5$$

$$\frac{\overline{u}}{u_r} = 2.44 \ln \frac{u_r y}{v} + 4.9 \text{ (turbulent region)} \qquad 30 < \frac{u_r y}{v}, \frac{y}{r_0} < 0.15$$

Dimensionless velocity distribution in the wall region for a smooth pipe

Velocity Profile



e = Average wall roughness height

 δ_v = Viscous wall layer thickness

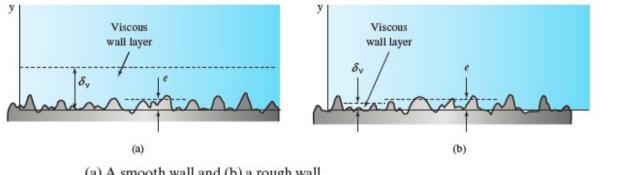
(a) A smooth wall and (b) a rough wall.

- For rough pipes, the viscous wall layer doesn't play an important role.
 - Turbulence starts from the protruding wall elements.
- Characteristic length is the Average roughness height e

$$\frac{\overline{u}}{u_r} = 2.44 \ln \frac{y}{e} + 8.5 \qquad \frac{y}{r_0} < 0.15$$

Dimensionless velocity profile for the wall region of a rough pipe

Velocity Profile



e = Average wall roughness height

 δ_v = Viscous wall layer thickness

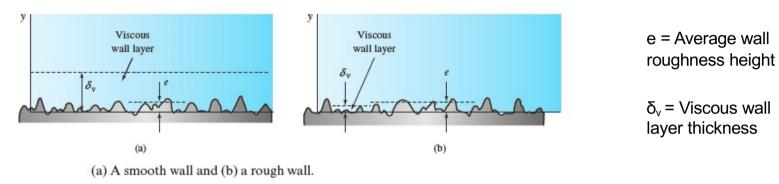
(a) A smooth wall and (b) a rough wall.

In the outer region, characteristic length is r₀ •

$$\frac{u_{\max} - \overline{u}}{u_{\tau}} = 2.44 \ln \frac{r_0}{y} + 0.8 \frac{y}{r_0} \le 0.15 \quad \text{(outer region)}$$

Velocity defect $(u_{max} - \bar{u})$ is normalized with $u_{\tau_{\cdot}}$ Relation is for both smooth and rough pipes.

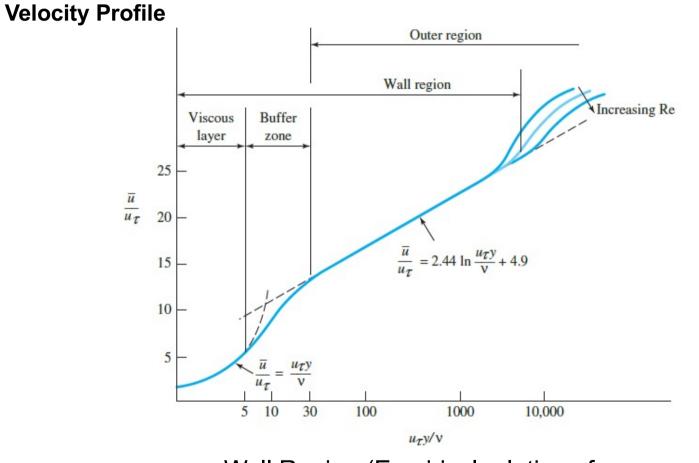




• The wall and outer regions may overlap. The maximum velocity is:

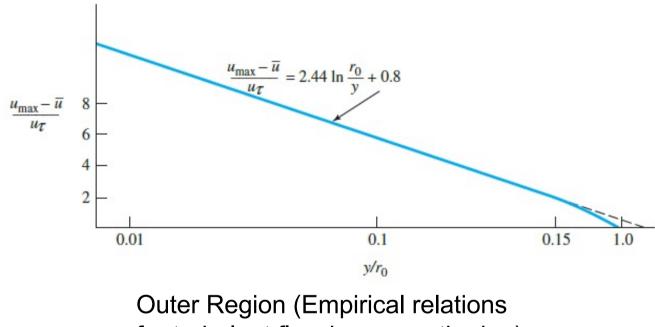
$$\frac{u_{\text{max}}}{u_r} = 2.44 \ln \frac{u_r r_0}{v} + 5.7 \qquad \text{(smooth pipes)}$$

$$\frac{u_{\text{max}}}{u_r} = 2.44 \ln \frac{r_0}{e} + 9.3 \qquad \text{(rough pipes)}$$

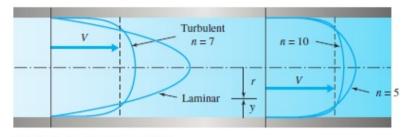


Wall Region (Empirical relations for turbulent flow in a smooth pipe)

Velocity Profile



for turbulent flow in a smooth pipe)



Turbulent velocity profile.

- The *power-law profile* describes the turbulent flow velocity distribution in a pipe:
 - Simpler form

$$\frac{\overline{u}}{u_{\max}} = \left(\frac{y}{r_0}\right)^{1/n}$$

Average velocity is then calculated to be:

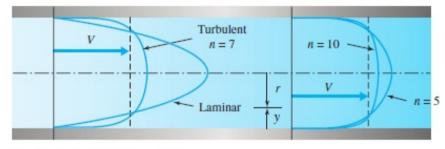
$$V = \frac{1}{\pi r_0^2} \int_0^{r_0} \overline{u}(r) 2\pi r \, dr = \frac{2n^2}{(n+1)(2n+1)} u_{\rm max}$$

 n: Depends on the friction factor f (Reynolds number and pipe wall roughness)

 Image: Table 7.1 Exponent n for Smooth Pipes

 \overline{f} Re = VD/v
 4 × 10³
 10⁵
 10⁶
 > 2 × 10⁶

 n
 6
 7
 9
 10



Turbulent velocity profile.

- The *power-law profile* for turbulent velocity flow distribution:
 - Cannot be used to obtain the slope at the wall (infinite for all n).
 - Cannot be used to predict wall shear stress.

Water at 20°C flows in a 100-mm-diameter pipe at an average velocity of 1.6 m/s. If the roughness elements are 0.046 mm high, would the wall be rough or smooth? Refer to Figure 7.10.

Solution

To determine if the wall is rough or smooth, we must compare the viscous wall layer thickness with the height of the roughness elements. So, let's find the viscous wall layer thickness. From Figure 7.11 the viscous layer thickness is determined by letting $u_r y/v = 5$, where $y = \delta_v$. First, we must find u_r . The Reynolds number is

$$Re = \frac{VD}{v} = \frac{1.6 \times 0.1}{10^{-6}} = 1.6 \times 10^{5}$$

From Table 7.1 $n \approx$ 7.5, so that, from Eq. 7.6.21,

$$f = \frac{1}{n^2} = \frac{1}{7.5^2} = 0.013$$

The wall shear is calculated from Eq. 7.3.19:

$$\tau_0 = \frac{1}{8}\rho V^2 f$$

= $\frac{1}{8} \times 1000 \times 1.6^2 \times 0.018 = 5.8 \text{ Pa}$

The friction velocity is found from the definition of the shear velocity:

$$u_{\tau} = \sqrt{\frac{\tau_0}{\rho}} = \sqrt{\frac{5.8}{1000}} = 0.076 \text{ m/s}$$

This allows us to calculate the viscous wall layer thickness using $y = \delta_y$:

$$\delta_{\nu} = \frac{5\nu}{u_{r}}$$
$$= \frac{5 \times 10^{-6} \text{ m}^{2}\text{/s}}{0.076 \text{ m/s}} = 6.6 \times 10^{-5} \text{ m} \text{ or } 0.066 \text{ mm}$$

Since the roughness elements are only 0.046 mm high, they are submerged in the viscous wall layer. Consequently, *the wall is smooth* (see Figure 7.10a). If the pipe were made of cast iron with e = 0.26 mm, the wall would be rough.

Note that the viscous wall layer, even at this relatively low velocity, is about 0.1% of the radius. The viscous wall layer is usually extremely thin.

The 40-mm-diameter smooth, horizontal pipe of Figure E7.10 transports 0.004 m³/s of water at 20°C. Using the power-law profile, approximate (a) the friction factor, (b) the maximum velocity, (c) the radial position where u = V, (d) the wall shear, (e) the pressure drop over a 10-m length, and (f) the maximum velocity using Eq. 7.6.16.

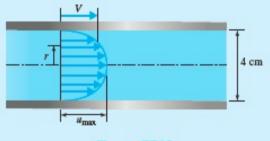


Figure E7.10

Solution

(a) The average velocity is calculated to be

$$V = \frac{Q}{A} = \frac{0.004}{\pi \times 0.02^2} = 3.18 \text{ m/s}$$

The Reynolds number is

$$Re = \frac{VD}{v} = \frac{3.18 \times 0.04}{10^{-6}} = 1.27 \times 10^{3}$$

From Table 7.1 we see that $n \cong 7.5$ and from Eq. 7.6.21,

$$f = \frac{1}{n^2} \\ = \frac{1}{7.5^2} = 0.018$$

(b) The maximum velocity is found using Eq. 7.6.20 to be

$$u_{\text{max}} = \frac{(n+1)(2n+1)}{2n^2}V$$
$$= \frac{8.5 \times 16}{2 \times 7.5^2} \times 3.18 = \underline{3.84 \text{ m/s}}$$

(c) The distance from the wall where u = V = 3.18 m/s is found using Eq. 7.6.19 as follows:

$$\frac{\overline{u}}{u_{\text{max}}} = \left(\frac{y}{r_0}\right)^{17.5}$$
$$\therefore y = r_0 \left(\frac{u}{u_{\text{max}}}\right)^{7.5}$$
$$= 2 \left(\frac{3.18}{3.84}\right)^{7.5} = 4.9 \text{ mm}$$

The radial position is thus

$$r = r_0 - y$$

= 20 - 4.9 = 15.1 mm

(d) The wall shear is found using Eq. 7.3.19 and is

$$\begin{aligned} \tau_0 &= \frac{1}{8} \rho V^2 f \\ &= \frac{1}{8} \times 1000 \times 3.18^2 \times 0.018 = \underline{23 \text{ Pa}} \end{aligned}$$

(e) The pressure drop is calculated using Eq. 7.6.18 with $\Delta p/L = -dp/dx$ to be

$$\Delta p = \frac{2\tau_0 L}{r_0} = \frac{2 \times 23 \times 10}{0.02} = 23\,000\,\text{Pa} \text{ or } \frac{23\,\text{kPa}}{23\,\text{kPa}}$$

(f) To use Eq. 7.6.16 we must know the shear velocity. It is

$$u_{\tau} = \sqrt{\frac{\tau_0}{\rho}}$$

= $\sqrt{\frac{23}{1000}} = 0.152 \text{ m/s}$

We then find u_{max} to be, using $v = 10^{-6} \text{ m}^2/\text{s}$,

$$u_{\text{max}} = 0.152 \left(2.44 \ln \frac{0.152 \times 0.02}{10^{-6}} + 5.7 \right) = \frac{3.84 \text{ m/s}}{3.84 \text{ m/s}}$$

This is the same as that given by the power-law formula in part (b). This answer is considered to be more accurate if it differs from that of Eq. 7.6.20. Note that the experimental data do not allow for accuracy in excess of three significant digits, and often to only two significant digits.

7.6.3 Losses in Developed Pipe Flow

- Most calculated quantity in pipe flow is the **head loss**.
 - Allows pressure change to be found → pump selection

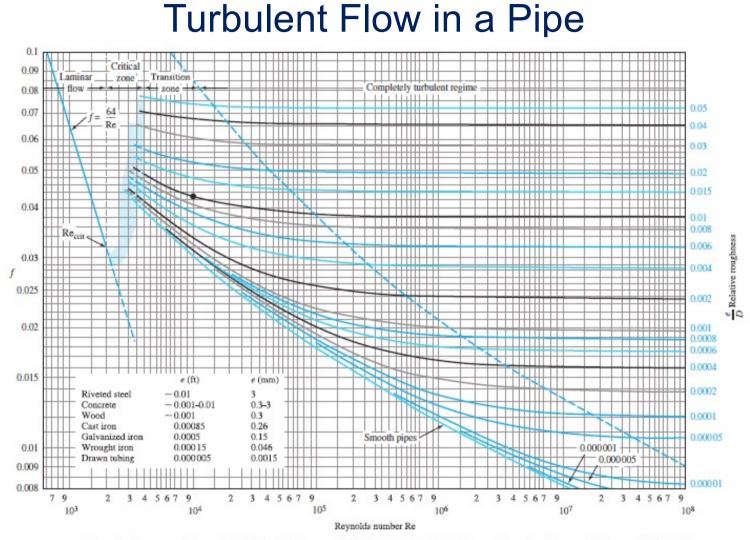
ection.
$$h_L = \frac{\Delta(p + \gamma h)}{\gamma}$$

• Derived from energy equation.

$$h_L = f \frac{L}{D} \frac{V^2}{2g}$$

Head loss from wall shear in a developed flow is related to the friction factor(f).

Darcy-Weisbach equation



Moody diagram. (From L. F. Moody, Trans. ASME, Vol. 66, 1944. Reproduced with permission of ASME.) (Note: If e/D = 0.01 and Re = 10⁴, the dot locates f = 0.043.)

Losses in Developed Pipe Flow

- Moody diagram is a plot of experimental data relating friction factor to the Reynolds number.
 - For fully developed pipe flow over a range of wall roughnesses.
- For a given wall roughness → There is a large enough Re to get a constant friction factor → Completely turbulent regime.
- For smaller relative roughness → As Re decreases, friction factor increases → Transition zone → Friction factor becomes like that of a smooth pipe.
- For Re < 2000→ The critical zone couples the turbulent flow to the laminar flow and may represent an oscillatory flow that alternately exists between turbulent and laminar flow.
- Assume new pipes → As a pipe gets older, corrosion occurs changing both the roughness and the pipe diameter.

Losses in Developed Pipe Flow

Smooth pipe flow: $\frac{1}{\sqrt{f}} = 0.86 \ln \operatorname{Re}\sqrt{f} - 0.8$ Empirical equations for Re > 4000

Completely turbulent zone:
$$\frac{1}{\sqrt{f}} = -0.86 \ln \frac{e}{3.7D}$$

Transition zone:
$$\frac{1}{\sqrt{f}} = -0.86 \ln\left(\frac{e}{3.7D} + \frac{2.51}{\text{Re}\sqrt{f}}\right)$$

- **Colebrook equation**: The equation that couples the smooth pipe equation • to the completely turbulent regime equation.
- Smooth pipe flow: e = 0; Completely turbulent zone: $Re = \infty$ •

Losses in Developed Pipe Flow

Category	Known	Unknown	
1	Q, D, e, v	h_L	
2	D, e, v, h_L	Q	
3	Q, e, v, h_L	D	

- Three types of problems for developed turbulent flow in a pipe:
 - One: Straightforward. Needs no iteration when using the Moody diagram.
 - Two and Three: Engineering design situation. Needs an iterative trial-and-error process.

Losses in Developed Pipe Flow

To avoid trial-and-error and the Moody diagram, use:

$$h_{L} = 1.07 \frac{Q^{2}L}{gD^{5}} \left\{ \ln \left[\frac{e}{3.7D} + 4.62 \left(\frac{vD}{Q} \right)^{0.9} \right] \right\}^{-2} \qquad 10^{-6} < e/D < 10^{-2} \\ 3000 < \text{Re} < 3 \times 10^{8} \\ Q = -0.965 \left(\frac{gD^{5}h_{L}}{L} \right)^{0.5} \ln \left[\frac{e}{3.7D} + \left(\frac{3.17v^{2}L}{gD^{3}h_{L}} \right)^{0.5} \right] \qquad \text{Re} > 2000 \\ D = 0.66 \left[e^{1.25} \left(\frac{LQ^{2}}{gh_{L}} \right)^{4.75} + vQ^{9.4} \left(\frac{L}{gh_{L}} \right)^{5.2} \right]^{0.04} \qquad 10^{-6} < e/D < 10^{-2} \\ 5000 < \text{Re} < 3 \times 10^{8} \\ \end{array}$$

- Developed by Swamee and Jain (1976) for pipe flow.
 - First and last equations are accurate to within 2% of the Moody diagram. The middle equation is as accurate as the Moody diagram.

Water at 20°C is transported for 450 m in a 38-mm-diameter wrought iron horizontal pipe with a flow rate of 2.75 L/s. Calculate the head loss and the pressure drop over the 450 m length of pipe, using (a) the Moody diagram and (b) the alternate method.

Solution

(a) The average velocity is

$$V = \frac{Q}{A} = \frac{2.75 \times 10^{-3}}{\pi \times (0.019)^2} = 2.43 \,\mathrm{m/s}$$

The Reynolds number is

$$\operatorname{Re} = \frac{VD}{V} = \frac{2.43 \times 0.038}{10^{-6}} = 92\,300$$

Obtaining *e* from Figure 7.13, we have, using D = 0.038 m,

$$\frac{e}{D} = \frac{0.000046}{0.038} = 0.0012$$

The friction factor is read from the Moody diagram to be

$$f = 0.023$$

The head loss is calculated as

$$h_{L} = f \frac{L}{D} \frac{V^{2}}{2g}$$
$$= 0.023 \frac{450}{0.038} \frac{(2.43)^{2}}{2 \times 9.8} = \underline{82 \text{ m}}$$

This answer is given to two significant numbers since the friction factor is known to at most two significant numbers. The pressure drop is found by Eq. 7.6.22 to be

$$\Delta p = \gamma h_L$$

= 9810 N/m³ × 82 = 804420 N/m² or 804.4 kPa

(b) The alternate method for this Category 1 problem uses Eq. 7.6.29, with D = 0.038 m:

$$h_{L} = 1.07 \frac{0.00275^{2} \times 450}{9.81 \times 0.038^{5}} \left\{ \ln \left[\frac{0.0012}{3.7} + 4.62 \left(\frac{10^{-6} \times 0.038}{0.00275} \right)^{0.9} \right]^{2} \right\}$$

= 82 m

This much simpler method provides the same value as that found using the Moody diagram.

A pressure drop of 700 kPa is measured over a 300-m length of horizontal, 100-mmdiameter wrought iron pipe that transports oil (S = 0.9, $v = 10^{-5}$ m²/s). Calculate the flow rate using (a) the Moody diagram, and (b) the alternate method.

Solution

(a) The relative roughness is

$$\frac{e}{D} = \frac{0.046}{100} = 0.00046$$

Assuming that the flow is completely turbulent (Re is not needed), the Moody diagram gives

f = 0.0165

The head loss is found to be

$$h_L = \frac{\Delta p}{\gamma_{\rm oil}} = \frac{700\ 000\ {\rm N/m^2}}{9800\ {\rm N/m^3} \times 0.9} = 79.4\ {\rm m}$$

The velocity is calculated from Eq. 7.6.23 to be

$$V = \left(\frac{2gDh_L}{fL}\right)^{1/2} = \left(\frac{2 \times 9.8 \text{ m/s}^2 \times 0.1 \text{ m} \times 79.4 \text{ m}}{0.0165 \times 300 \text{ m}}\right)^{1/2} = 5.61 \text{ m/s}$$

This provides us with a Reynolds number of

$$Re = \frac{VD}{v} = \frac{5.61 \text{ m/s} \times 0.1 \text{ m}}{10^{-5} \text{ m}^2/\text{s}} = 5.61 \times 10^4$$

Using this Reynolds number and e/D = 0.00046, the Moody diagram gives the friction factor as

f = 0.023

This corrects the original value for f. The velocity is recalculated to be

$$V = \left(\frac{2 \times 9.8 \times 0.1 \times 79.4}{0.023 \times 300}\right)^{1/2} = 4.75 \text{ m/s}$$

The Reynolds number is then

$$Re = \frac{4.75 \times 0.1}{10^{-5}} = 4.75 \times 10^{4}$$

From the Moody diagram f = 0.023 appears to be satisfactory. Thus the flow rate is

$$Q = VA = 4.75 \times \pi \times 0.05^2 = 0.037 \text{ m}^3\text{/s}$$

Only two significant numbers are given since f is known to at most two significant numbers. (b) The alternative method for this Category 2 problem uses the explicit relationship (7.6.30). We can directly calculate Q to be

$$Q = -0.965 \left(\frac{9.8 \times 0.1^5 \times 79.4}{300}\right)^{0.5} \ln \left[\frac{0.00046}{3.7} + \left(\frac{3.17 \times 10^{-10} \times 300}{9.8 \times 0.1^3 \times 79.4}\right)^{0.5}\right]$$
$$= -0.965 \times 5.096 \times 10^{-3} \times (-7.655) = 0.038 \text{ m}^3/\text{s}$$

This much simpler method produces a value essentially the same as that obtained using the Moody diagram.

Drawn tubing of what diameter should be selected to transport 0.002 m³/s of 20°C water over a 400-m length so that the head loss does not exceed 30 m? (a) Use the Moody diagram and (b) the alternative method.

Solution

(a) In this problem we do not know D. Thus, a trial-and-error solution is anticipated. The average velocity is related to D by

$$V = \frac{Q}{A} = \frac{0.002}{\pi D^2/4} = \frac{0.00255}{D^2}$$

The friction factor and D are related as follows:

$$h_{L} = f \frac{L}{D} \frac{V^{2}}{2g}$$

$$30 = f \frac{400}{D} \frac{(0.00255/D^{2})}{2 \times 9.8}$$

$$D^{5} = 4.42 \times 10^{-6} f$$

The Reynolds number is

$$\operatorname{Re} = \frac{VD}{v} = \frac{0.00255D}{D^2 \times 10^{-6}} = \frac{2550}{D}$$

Now, let us simply guess a value for f and check with the relations above and the Moody diagram. The first guess is f = 0.03, and the correction is listed in the following table. Note: the second guess is the value for f found from the calculations of the first guess.

f	<i>D</i> (m)	Re	elD	f(Figure 7.13)
0.03	0.0421	6.06×10^{4}	0.000036	0.02
0.02	0.0388	6.57×10^{4}	0.000039	0.02

The value of f = 0.02 is acceptable, yielding a diameter of 38.8 mm. Since this diameter would undoubtedly not be standard, a diameter of

$$D = 40 \text{ mm}$$

would be the tube size selected. This tube would have a head loss less than the limit of $h_L = 30$ m imposed in the problem statement. Any larger-diameter tube would also satisfy this criterion but would be more costly, so it should not be selected.

(b) The alternative method for this Category 3 problem uses the explicit relationship (7.6.31). We can directly calculate D to be

$$D = 0.66 \left[(1.5 \times 10^{-6})^{1.25} \left(\frac{400 \times 0.002^2}{9.81 \times 30} \right)^{4.75} + 10^{-6} \times 0.002^{9.4} \left(\frac{400}{9.81 \times 30} \right)^{5.2} \right]^{0.04}$$

= 0.66 [5.163×10⁻³³ + 2.102 × 10⁻³¹]^{0.04} = 0.039 m

Hence D = 40 mm would be the tube size selected. This is the same tube size as that selected using the Moody diagram.

Losses in Noncircular Conduits

- Can approximate for conduits with noncircular cross sections:
 - Using hydraulic radius R

$$R = \frac{A}{P}$$

A: Cross-sectional area

P: Wetted perimeter → Perimeter where the fluid is in contact with the solid boundary

- E.g., for a circular pipe:
 - Hydraulic radius $R = r_o/2$

$$Re = \frac{4VR}{v}$$
 relative $= \frac{e}{4R}$

• The head-loss then becomes:

$$h_L = f \frac{L}{4R} \frac{V^2}{2g}$$

Air at standard conditions is to be transported through 500 m of a smooth, horizontal, $300 \text{ mm} \times 200 \text{ mm}$ rectangular duct at a flow rate of 0.24 m³/s. Calculate the pressure drop.

Solution

The hydraulic radius is

$$R = \frac{A}{P} = \frac{0.3 \times 0.2}{(0.3 + 0.2) \times 2} = 0.06 \text{ m}$$

The average velocity is

$$V = \frac{Q}{A} = \frac{0.24}{0.3 \times 0.2} = 4.0 \text{ m/s}$$

This gives a Reynolds number of

$$\operatorname{Re} = \frac{4VR}{v} = \frac{4 \times 4 \times 0.06}{1.5 \times 10^{-5}} = 6.4 \times 10^{4}$$

Using the smooth pipe curve of the Moody diagram, there results

$$f = 0.0196$$

Hence,

$$h_L = f \frac{L}{4R} \frac{V^2}{2g} = 0.0196 \frac{500 \text{ m}}{4 \times 0.06 \text{ m}} \frac{4^2 \text{ m}^2/\text{s}^2}{2 \times 9.8 \text{ m/s}^2} = 33.3 \text{ m}$$

The pressure drop is

$$\Delta p = \rho g h_L = 1.23 \times 9.8 \times 33.3 = \underline{402} \text{ Pa}$$

Minor Losses in Pipe Flow

- Sometimes *minor losses* (from fittings that cause additional losses) can exceed frictional losses.
- Expressed in terms of a loss coefficient K.

$$h_L = K \frac{V^2}{2g}$$

- K can be determined experimentally.
 - If there is an expansion from area A_1 to area A_2 :

$$h_L = \left(1 - \frac{A_1}{A_2}\right)^2 \frac{V_1^2}{2g}$$

• For a sudden expansion in area:

$$K = \left(1 - \frac{A_1}{A_2}\right)^2$$

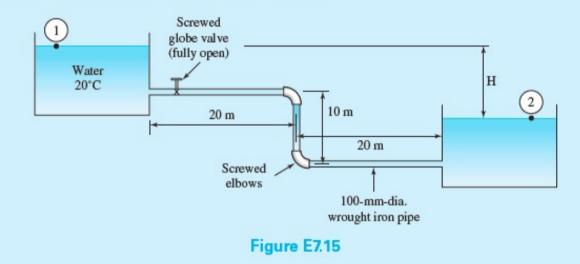
Minor Losses in Pipe Flow

• A loss coefficient can be expressed as an *equivalent length L_e* of pipe:

$$L_e = K \frac{D}{f}$$

• For long segments of pipe, minor losses can usually be neglected.

If the flow rate through a 100-mm-diameter wrought iron pipe (Figure E7.15) is 0.04 m³/s, find the difference in elevation H of the two reservoirs.



Solution

The energy equation written for a control volume that contains the two reservoir surfaces (see Eq. 4.5.17), where $V_1 = V_2 = 0$ and $p_1 = p_2 = 0$, is

$$0 = z_2 - z_1 + h_L$$

Thus, letting $z_1 - z_2 = H$, we have

$$H = \left(K_{\text{entrance}} + K_{\text{valve}} + 2K_{\text{elbow}} + K_{\text{exit}}\right) \frac{V^2}{2g} + f \frac{L}{D} \frac{V^2}{2g}$$

The average velocity, Reynolds number, and relative roughness are

$$V = \frac{Q}{A} = \frac{0.04}{\pi \times 0.05^2} = 5.09 \text{ m/s}$$

Re = $\frac{VD}{v} = \frac{5.09 \times 0.1}{10^{-6}} = 5.09 \times 10^5$
 $\frac{e}{D} = \frac{0.046}{100} = 0.00046$

From the Moody diagram we find that

f = 0.0173

Using the loss coefficients from Table 7.2 for an entrance, a globe valve, screwed 10-cm-diameter standard elbows, and an exit there results

$$H = (0.5 + 5.7 + 2 \times 0.64 + 1.0) \frac{5.09^2}{2 \times 9.8} + 0.0173 \frac{50}{0.1} \frac{5.09^2}{2 \times 9.8}$$

= 11.2 + 11.4 = 22.6 m

Note: The minor losses are about equal to the frictional losses as expected, since there are five minor loss elements in 500 diameters of pipe length.

Approximate the loss coefficient for the sudden contraction $A_1/A_2 = 2$ by neglecting the losses in the contracting portion up to the vena contracta and assuming that all the losses occur in the expansion from the vena contracta to A_2 (see Figure 7.16). Compare with that given in Table 7.2.

Solution

The head loss from the vena contracta to area A_2 is (see Table 7.2, sudden enlargement)

$$h_L = \left(1 - \frac{A_e}{A_2}\right)^2 \frac{V_e^2}{2g}$$

Continuity allows us to write (V_c is the velocity at the area A_c)

$$V_c = \frac{A_2}{A_c} V_2$$

Thus, the head loss based on V_2 is

$$h_L = \left(1 - \frac{A_e}{A_2}\right)^2 \left(\frac{A_2}{A_e}\right)^2 \frac{V_2^2}{2g}$$

so the loss coefficient of Eq. 7.6.35 is

$$K = \left(1 - \frac{A_c}{A_2}\right)^2 \left(\frac{A_2}{A_c}\right)^2$$

Using the expression of C_c given in Figure 7.16, we have

$$\frac{A_c}{A_2} = C_c = 0.62 + 0.38 \left(\frac{1}{2}\right)^3 = 0.67$$

Finally,

$$K = (1 - 0.67)^2 \frac{1}{0.67^2} = 0.24$$

This compares favorably with the value of 0.25 given in Table 7.2.

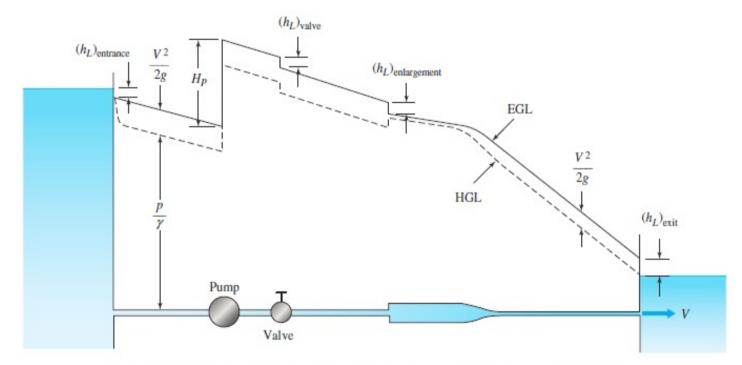
Hydraulic and Energy Grade Lines

 Energy equation is written in a form where the terms have dimensions of length:

$$-\frac{\dot{W_s}}{\dot{m}g} - \frac{V_2^2 - V_1^2}{2g} + \frac{p_2 - p_1}{\gamma} + z_2 - z_1 + h_L$$

- The Hydraulic and Energy grade lines for piping systems can hence be defined:
 - **Hydraulic Grade Line (HGL):** Located a distance p/γ above the center of the pipe.
 - Energy Grade Line (EFL): Located a distance V²/2g above the HGL.

Hydraulic and Energy Grade Lines

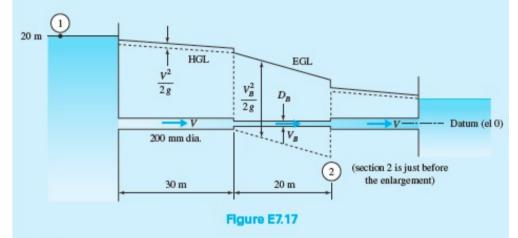


Hydraulic grade line (HGL) and energy grade line (EGL) for a piping system.

Hydraulic and Energy Grade Lines: NOTES

- As V→0, HGL and EGL approach each other. (For a reservoir, they are identical and lie on the surface.)
- EGL and HGL slope downward in the flow direction (due to head loss).
 - Greater the loss per unit length, greater the slope.
- HGL and EGL suddenly change when a loss occurs due to sudden geometry changes.
 - Jumps when useful energy is added (pump).
 - Drops when useful energy is extracted (turbine).
- If the HGL passes through the centerline of the pipe \rightarrow pressure is zero.
 - If the pipe is above the HGL \rightarrow vacuum condition.

Water at 20°C flows between two reservoirs at the rate of 0.06 m³/s as shown in Figure E7.17. Sketch the HGL and the EGL. What is the minimum diameter D_{g} allowed to avoid the occurrence of cavitation?



Solution

The EGL and the HGL are sketched on the figure, including sudden changes at the entrance, contraction, enlargement, and the exit. Note the large velocity head (the difference between the EGL and the HGL) in the smaller pipe because of the high velocity. The velocity, Reynolds number, and relative roughness in the 20-cm-diameter pipe are calculated to be

$$V = \frac{Q}{A} = \frac{0.06}{\pi \times 0.20^2/4} = 1.91 \text{ m/s}$$

Re = $\frac{VD}{v} = \frac{1.91 \times 0.2}{10^{-6}} = 3.8 \times 10^5$
 $\frac{e}{D} = \frac{0.26}{200} = 0.0013$

Thus f = 0.022 from Figure 7.13. The velocity, Reynolds number, and relative roughness in the smaller pipe are

$$V_{B} = \frac{0.06}{\pi D_{B}^{2}/4} = \frac{0.0764}{D_{B}^{2}}$$
$$Re_{B} = \frac{0.0764 \times D_{B}}{D_{B}^{2} \times 10^{-6}} = \frac{76400}{D_{B}}$$
$$\frac{e}{D_{B}} = \frac{0.00026}{D_{B}}$$

The minimum possible diameter is established by recognizing that the water vapor pressure (2450 Pa absolute) at 20°C is the minimum allowable pressure. Since the distance between the pipe and the HGL is an indication of the pressure in the pipe, we can conclude that the minimum pressure will occur at section 2. Hence the energy equation applied between section 1, the reservoir surface, and section 2 gives

$$\frac{V_{f}}{2g}^{0} + \frac{p_{1}}{\gamma} + z_{1} = \frac{V_{B}^{2}}{2g} + \frac{p_{2}}{\gamma} + z_{2}^{0} + K_{ent}\frac{V_{A}^{2}}{2g} + K_{oont}\frac{V_{B}^{2}}{2g} + f_{A}\frac{L_{A}}{D_{A}}\frac{V_{A}^{2}}{2g} + f_{B}\frac{L_{B}}{D_{B}}\frac{V_{B}^{2}}{2g}$$

where the subscript A refers to the 20-cm-diameter pipe. This simplifies, using absolute pressure, to

$$\frac{101\ 000}{9810} + 20 = \frac{\left(0.0764/D_B^2\right)^2}{2 \times 9.81} \left(1 + 0.25 + f_B \frac{20}{D_B}\right) + \frac{2450}{9810} + \left(0.5 + 0.022\frac{30}{0.2}\right)\frac{1.91^2}{2 \times 9.81}$$

$$98\ 600 = \frac{1.25}{D_B^4} + f_B \frac{20}{D_B^5}$$

where we have used $K_{ent} = 0.5$ and assumed that $K_{eout} = 0.25$. This requires a trial-anderror solution. The following illustrates the procedure.

Let $D_B = 0.1$ m. Then $e/D_B = 0.0026$ and $\text{Re}_B = 7.6 \times 10^5$. Therefore, f = 0.026:

98 600 2 12 500 + 52 000

Let $D_B = 0.09$ m. Then $e/D_B = 0.0029$ and $\text{Re}_B = 8.4 \times 10^5$. Therefore, f = 0.027:

98 600 2 19 000 + 91 000

We see that 0.1 m is too large and 0.09 m is too small. In fact, the value of 0.09 m is only slightly too small. Consequently, to be safe we must select the next larger pipe size of 0.1 m diameter. If there were a pipe size of 95 mm diameter, that could be selected. Assuming that that size is not available, we select

$$D_B = 100 \text{ mm}$$

Note that the assumption of a 2:1 area ratio for the contraction is too small. It is actually 4:1. This would give $K_{\text{cont}} \simeq 0.4$. After a quick check we conclude that this value does not significantly influence the result.

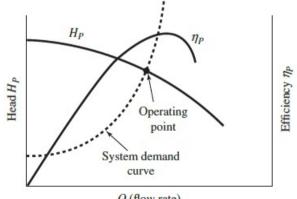
Simple Pipe System with a Centrifugal Pump

- If the flow rate of the pump is not given \rightarrow Not straightforward.
 - The head produced by the pump and the efficiency depend on the discharge.
 - Need the characteristic curves of the pump.
 - Can relate flow rate Q, and pump head H_P.

 $H_P = c_1 + c_2 Q^2$

System Demand Curve: Energy equation relating pump head to an unknown

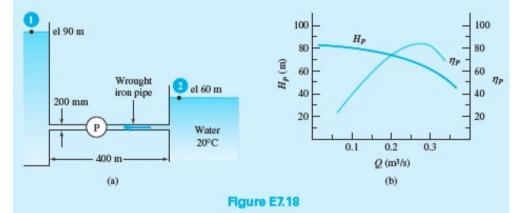
flow rate.



Q (flow rate)

Pump characteristic curve (Intersection is the operating point of the system)

Estimate the flow rate in the simple piping system of Figure E7.18a if the pump characteristic curves are as shown in Figure E7.18b. Also, find the pump power requirement.



Solution

We will assume that the Reynolds number is sufficiently large that the flow is completely turbulent. So, using e/D = 0.046/200 = 0.00023, the friction factor from the Moody diagram is

$$f = 0.014$$

The energy equation (see Eq. 7.6.40), with $H_r = -\dot{W_r}/\dot{mg}$, applied between the two surfaces, yields

$$H_r = \frac{V_2^2 \not Z_1^p}{2g} + z_2 - z_1 + \frac{p_2 \not Z_1^p}{\gamma} + h_L$$

or

$$H_r = 90 - 60 + \left(K_{\text{entrance}} + K_{\text{exit}} + f\frac{L}{D}\right)\frac{V^2}{2g}$$

$$= 30 + \left(0.5 + 1.0 + 0.014 \frac{400}{0.2}\right) \frac{Q^2}{2 \times 9.8 \times [\pi \times 0.1^2]^2}$$
$$= 30 + 1520Q^2$$

This equation, the system demand curve, and the characteristic curve $H_r(Q)$ of the pump are now solved simultaneously by trial and error. Actually, the curve could be plotted on the same graph as the characteristic curve, and the point of intersection, the operating point, would provide Q. Try $Q = 0.2 \text{ m}^3/\text{s}$: $(H_r)_{\text{energy}} = 91 \text{ m}$, $(H_r)_{\text{char}} \cong 75 \text{ m}$. Try $Q = 0.15 \text{ m}^3/\text{s}$: $(H_r)_{\text{energy}} = 64 \text{ m}$, $(H_r)_{\text{char}} \cong 75 \text{ m}$. Try $Q = 0.17 \text{ m}^3/\text{s}$: $(H_r)_{\text{energy}} = 74 \text{ m}$, $(H_r)_{\text{char}} \cong 76 \text{ m}$. This is our solution. We have

$$Q = 0.17 \text{ m}^3/\text{s}$$

Check the Reynolds number: Re = $DQ/Av = 0.2 \times 0.17/(\pi \times 0.1^2 \times 10^{-6}) = 1.08 \times 10^6$. This is sufficiently large, but marginally so.

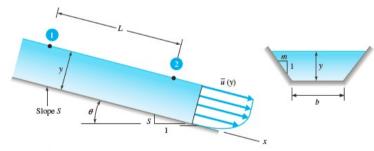
The power requirement of the pump is given by Eq. 4.5.26:

$$\dot{W_r} = \frac{Q\gamma H_r}{\eta_r} = \frac{0.17 \text{ m}^3\text{/s} \times 9800 \text{ N/m}^3 \times 75 \text{ m}}{0.65} = 198\ 000 \text{ W} \text{ or } 198 \text{ kW}$$

where the efficiency $\eta_r = 0.65$ is found from the characteristic curve at Q = 0.17 m³/s. Note: Since L/D > 1000, minor losses due to the entrance and exit could have been neglected.

- Steady, uniform flow in an open channel can be understood using the Darcy-Weisbach relation.
 - Uniform flow in an open, rough channel can be analyzed using the energy equation.

$$0 = \frac{V_2^2 - V_1^2}{2g} + \frac{P_2 - P_1}{\gamma} + z_2 - z_1 + h_z$$



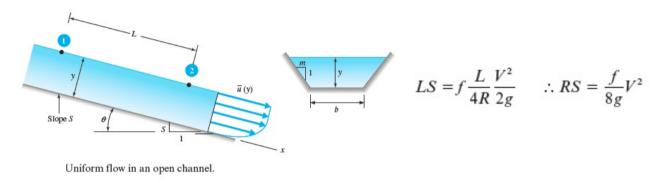
Uniform flow in an open channel.

• Hence the head loss is simply:

 $h_L = z_1 - z_2$ L: Length of the channel = $L \sin \theta = LS$ S: Slope of the channel

 The Darcy-Weisbach equation for this headloss is:

$$LS = f \frac{L}{4R} \frac{V^2}{2g}$$
 $\therefore RS = \frac{f}{8g} V^2$ R: Hydraulic radius



For large open channels (having large Reynolds numbers), the friction factor is in • the turbulent region.

 $V = C\sqrt{RS}$

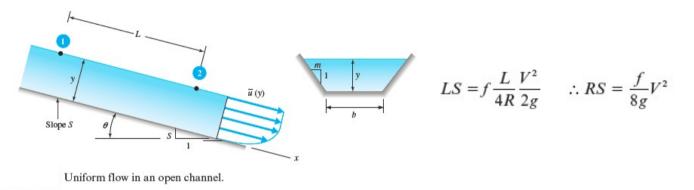
C = Chezy coefficient (dimensional • constant)

$$C = \frac{c_1}{n} R^{1/6}$$

- C = Channel roughness .
- R = Hydraulic radius •

n: Dimensionless constant related to the wall . roughness (Manning n)

Wall material	Manning n
Planed wood	0.012
Unplaned wood	0.013
Finished concrete	0.012
Unfinished concrete	0.014
Sewer pipe	0.013
Brick	0.016
Cast iron, wrought iron	0.015
Concrete pipe	0.015
Riveted steel	0.017
Earth, straight	0.022
Corrugated metal flumes	0.025
Rubble	0.03
Earth with stones and weeds	0.035
Mountain streams	0.05



• The flow-rate in an open channel is found using the **Chezy-Manning equation**.

Wall material	Manning 1
Planed wood	0.012
Unplaned wood	0.013
Finished concrete	0.012
Unfinished concrete	0.014
Sewer pipe	0.013
Brick	0.016
Cast iron, wrought iron	0.015
Concrete pipe	0.015
Riveted steel	0.017
Earth, straight	0.022
Corrugated metal flumes	0.025
Rubble	0.03
Earth with stones and weeds	0.035
Mountain streams	0.05

Note: Equation is usually used for rough-walled channels.

The depth of water at 16°C flowing in a 3.6-m-wide rectangular, finished concrete channel is measured to be 1.2 m. The slope is measured to be 0.0016. Estimate the flow rate using (a) the Chezy-Manning equation and (b) the Darcy-Weisbach equation.

Solution

The hydraulic radius is calculated to be

$$R = \frac{A}{P} = \frac{yb}{2y+b} = \frac{1.2 \times 3.6}{2 \times 1.2 + 3.6} = 0.72 \,\mathrm{m}$$

(a) Using the Chezy-Manning equation, with n = 0.012 from Table 7.3 and c = 1, we have

$$Q = \frac{1}{n} A R^{2/3} S^{1/2}$$

= $\frac{1 \text{ m}^{1/3}/\text{s}}{0.012} \times (1.2 \times 3.6) \text{ m}^2 \times (0.72)^{2/3} \text{ m}^{2/3} \times 0.0016^{1/2} = \underline{11.57 \text{ m}^3/\text{s}}$

(b) The relative roughness is, using a low value e = 0.00045 m (it is finished concrete) shown on the Moody diagram:

$$\frac{e}{4R} = \frac{0.00045}{4 \times 0.72} = 0.00016$$

Assuming a completely turbulent flow, the Moody diagram gives the friction factor as

f = 0.013

The Darcy-Weisbach equation (7.7.3) then yields the velocity as follows:

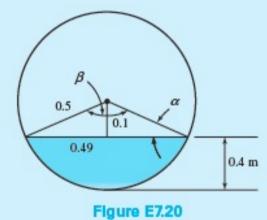
$$V = \left(\frac{8RgS}{f}\right)^{1/2}$$
$$= \left(\frac{8 \times 0.72 \text{ m} \times 9.81 \text{ m/s}^2 \times 0.0016}{0.013}\right)^{1/2} = 2.64 \text{ m/s}^2$$

The flow rate is calculated as

$$Q = VA = 2.64 \times 1.2 \times 3.6 = 11.4 \text{ m}^3\text{/s}$$

These two values are within 2%, an acceptable engineering tolerance for this type of problem. That found using the Moody diagram is considered to be more accurate, however.

A 1.0-m-diameter concrete pipe transports 20°C water at a depth of 0.4m. If the slope is 0.001, find the flow rate using (a) the Chezy-Manning equation and (b) the Darcy-Weisbach equation.



Solution

From the sketch of the pipe in Figure E7.20 the following are calculated:

$$\alpha = \sin^{-1} \frac{0.1}{0.5} = 11.54^{\circ}$$

$$\therefore \beta = 180 - 2 \times 11.54 = 156.9^{\circ}$$

$$\therefore A = \pi \times 0.5^{2} \times \frac{156.9}{360} - 0.49 \times 0.1 = 0.2933 \,\mathrm{m}^{2}$$

$$P = 2\pi \times 0.5 \times \frac{156.9}{360} = 1.369 \,\mathrm{m}$$

The hydraulic radius is found, using the above calculations, to be

$$R = \frac{A}{P} = \frac{0.2933}{1.369} = 0.2142 \,\mathrm{m}$$

(a) The Chezy-Manning equation yields, with n from Table 7.3 and $c_1 = 1.0 \text{ m}^{1/3}/\text{s}$,

$$Q = \frac{1.0}{n} A R^{2l_3} S^{1/2} = \frac{1.0}{0.015} \times 0.2933 \times 0.2142^{2l_3} \times 0.001^{1/2} = 0.22 \text{ m}^3/\text{s}$$

(b) The relative roughness is, using a relatively rough value for concrete pipe from Figure 7.13, as suggested by Table 7.3 of e = 20 mm,

$$\frac{e}{4R} = \frac{2}{4 \times 214.2} = 0.0023$$

Assuming completely turbulent flow, the Moody diagram yields

f = 0.025

The Darcy-Weisbach equation (7.7.3) then gives the following:

:.
$$V = \left(\frac{8RgS}{f}\right)^{1/2} = \left(\frac{8 \times 0.2142 \times 9.81 \times 0.001}{0.025}\right)^{1/2} = 0.820 \,\mathrm{m/s}$$

The flow rate is

$$Q = VA = 0.820 \times 0.2933 = 0.24 \text{ m}$$

This is within 8% of the result above, an acceptable tolerance for this type of problem. The second method, which is more difficult to apply, is considered to be more accurate, however.

Summary

• Laminar entrance lengths for a pipe and wide channel are:

$$\frac{L_E}{D} = 0.065 \text{ Re} \qquad \frac{L_E}{h} = 0.04 \text{ Re}$$

• For high Reynolds-number turbulent pipe flow, the entrance length is:

$$\frac{L_E}{D} = 120$$

• For laminar flow in a pipe and a wide channel, the pressure drop and friction factor are:

a: Channel height

$$\Delta p = \frac{8\mu VL}{r_0^2} \qquad f = \frac{64}{\text{Re}} \qquad \text{pipe}$$
$$\Delta p = \frac{12\mu VL}{a^2} \qquad f = \frac{48}{\text{Re}} \qquad \text{channel}$$

Summary

• The torque required to rotate an inner cylinder with the outer cylinder fixed is: $4\pi\mu r_1^2 r_2^2 L\omega_1$

$$T = \frac{4\pi\mu r_1^2 r_2^2 L\omega_1}{r_2^2 - r_1^2}$$

• The head loss in a turbulent flow is simply:

$$h_L = f \frac{L}{D} \frac{V^2}{2g}$$
 f: Found using a Moody diagram

• Minor losses are included using loss coefficients, K:

$$h_L = K \frac{V^2}{2g}$$

• The flow rate in an open channel is estimated by:

$$Q = \frac{c_1}{n} A R^{2/3} S^{1/2} \qquad c_1 = 1.0 \text{ m}^{1/3} \text{/s}$$